

SDN500

MEMS Integrated SAASM GPS-AJ / INS Tactical System

Ideal for High-Precision Navigation & Guidance Applications:

- Precision Guided Munitions
- Tactical Missiles
- Position Sensor for Geo-Surveying
- Targeting & Positioning
- Precision Antenna Pointing
- UAVs & Other Unmanned Vehicles
- Targets & Drones
- Ground Vehicle Tracking
- Range Instrumentation

Key Performance Features:

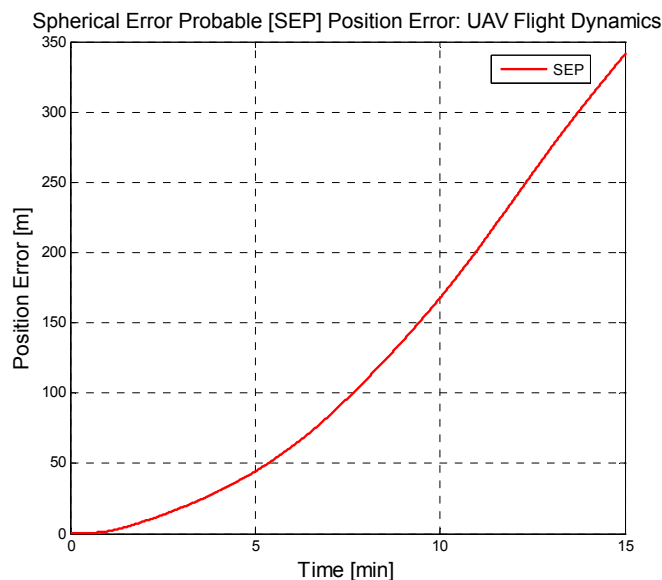
- **Position 3.9 m SEP**
- **Attitude**
 - Roll/Pitch (1σ) – 1.0 mrad (0.057°)
 - Heading in Dynamics (1σ) – 1.5 mrad (0.086°)
- **Integrates with External NavFire™ SAASM GPS Anti-Jam (AJ) Receiver**
 - 12 channels with all-in-view, L1 frequency
- **25 in.³ Compact Size for Packaging Flexibility**
- **Weighs <1.6 lbs.**
- **Customer Programmable Output Data Rates**



The SDN500 GPS-AJ/INS navigation system is a platform extension of Systron Donner Inertial's (SDI) proven tactical grade SDI500 IMU. The SDN500 GPS-AJ/INS combines latest generation quartz MEMS gyros and accelerometers, delivers industry leading bias in-run stability performance, provides enhanced 100Hz position data and integrates an external Rockwell Collins NavFire™ SAASM GPS Anti-Jam (AJ) Receiver, creating a tightly coupled powerful GPS/INS System for guidance and navigation applications. The modular compact 25 in³ size provides for maximum packaging flexibility in dense systems.

The solid state quartz sensors and sealed construction provide reliable 50,000+ hr. MTBF, and a 20 year operating and storage life. Continuous Built-in Test (BIT), configurable communications protocols, electromagnetic interference (EMI) protection, and flexible input power requirements make the SDN500 easy to use in a wide range of higher order integrated system applications.

	Units	Measure	SDN500-AD76	SDN500-BD76	SDN500-CD76
System Performance					
Position (SEP)	m	max		3.9	
Velocity (horizontal/vertical)	m/s	1 σ		0.1/0.1	
Pitch/Roll	mrad	1 σ		1.0	
Heading (in motion)	mrad	1 σ		1.5 + d ¹	
Timemark Output 1pps	μ s	nom		\pm 1	
Gyro Channels					
Bias In-Run Stability from Turn-on	deg/hr	1 σ	1.0	1.5	2.0
Angle Random Walk	deg/ \sqrt hr	1 σ	0.02	0.02	0.03
Angular Rate – Dynamic Range	deg/sec	min	\pm 1000	\pm 1000	\pm 1000
Accelerometer Channels					
Bias In-Run Stability from Turn-on	μ g	1 σ	100	200	200
Random Walk Noise	μ g/ \sqrt Hz	1 σ	100	100	120
Acceleration – Dynamic Range	g	min	\pm 50	\pm 50	\pm 50
System Physical & Environmental					
Input Voltage	Vdc			+12 to +42	
Power	watts			<7.5	
I/O				RS232/422, SDLC IMU Output	
Volume	cu in			25	
Weight	lbs			<1.6	
Temperature Range (Operating)	$^{\circ}$ C			-45 to +85	
Vibration (Operating)	g _{RMS}			12	
Shock (Operating)	g, msec			40, 30	
Altitude (INS/GPS)	ft			60,000	
Reliability @ 35 $^{\circ}$ C	hrs			50,000 MTBF, ground: 6,000 MTBF, air cargo	



¹ d represents a growth rate that depends on the time once all horizontal accelerations have stopped, drift will be 1 to 10 deg/hr 1 σ .

For more information, contact:

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